

Last Time

Chapter 2 Rigid Body Motion

1 Rigid Body Transformations

- Length Preserving: $\|g(p) - g(q)\| = \|p - q\|$
- Orientation Preserving: $g_*(v \times w) = g_*(v) \times g_*(w)$

2 Rotational motion in \mathbb{R}^3

- Rotation Matrix
 - Represents *configuration*
 - Represents (*rotational*) *transformation*
- Rotation Matrices with matrix multiplication form a *Group*
- Rotational Transformation is a *Rigid Body Transformation*

Recap

Today

Chapter 2 Rigid Body Motion

2 Rotational motion in \mathbb{R}^3

- The *Exponential Map*
- *Rodrigues Formula*
- *Euler Angles*
- *Quaternions*

2.2 Rotational Motion in \mathbb{R}^3

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□ Parametrization of $SO(3)$ (the exponential coordinate):

◇ **Review:** $S^1 = \{z \in \mathbb{C} \mid |z| = 1\}$

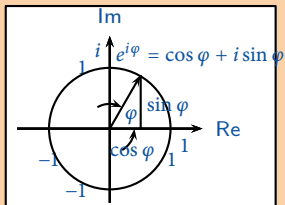


Figure 2.4

Euler's Formula

“One of the most remarkable, almost astounding, formulas in all of mathematics.”

R. Feynman

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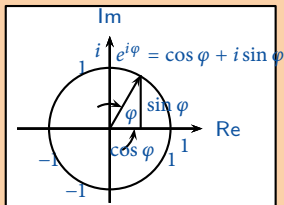


Figure 2.4

Euler's Formula

“One of the most remarkable, almost astounding, formulas in all of mathematics.”

R. Feynman

◇ **Review:**

$$\begin{cases} \dot{x}(t) = ax(t) \\ x(0) = x_0 \end{cases} \Rightarrow x(t) = e^{at}x_0$$

2.2 Rotational Motion in \mathbb{R}^3

$$R \in SO(3), R = \begin{bmatrix} r_{11} & r_{12} & r_{13} \\ r_{21} & r_{22} & r_{23} \\ r_{31} & r_{32} & r_{33} \end{bmatrix}$$

$$r_i \cdot r_j = \begin{cases} 0, & i \neq j \\ 1, & i = j \end{cases} \quad \leftarrow 6 \text{ constraints}$$

$\Rightarrow 3$ independent parameters!

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Consider motion of a point q on a rotating link

$$\begin{cases} \dot{q}(t) = \omega \times q(t) = \hat{\omega}q(t) \\ q(0): \text{Initial coordinates} \end{cases}$$

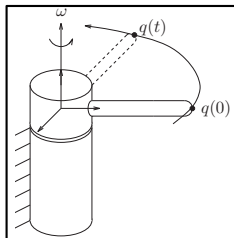


Figure 2.5

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$$\Rightarrow q(t) = e^{\hat{\omega}t} q_0 \text{ where } e^{\hat{\omega}t} = I + \hat{\omega}t + \frac{(\hat{\omega}t)^2}{2!} + \frac{(\hat{\omega}t)^3}{3!} + \dots$$

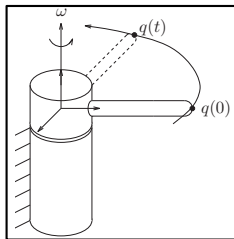


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2.2 Rotational Motion in \mathbb{R}^3

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Consider motion of a point q on a rotating link

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$$\Rightarrow q(t) = e^{\hat{\omega}t} q_0 \text{ where } e^{\hat{\omega}t} = I + \hat{\omega}t + \frac{(\hat{\omega}t)^2}{2!} + \frac{(\hat{\omega}t)^3}{3!} + \dots$$

By the definition of rigid transformation, $R(\omega, \theta) = e^{\hat{\omega}\theta}$. Let $so(3) = \{\hat{\omega} | \omega \in \mathbb{R}^3\}$ or $so(n) = \{S \in \mathbb{R}^{n \times n} | S^T = -S\}$ where $\wedge : \mathbb{R}^3 \mapsto so(3) : \omega \mapsto \hat{\omega}$, we have:

Property 3: $\exp : so(3) \mapsto SO(3), \hat{\omega}\theta \mapsto e^{\hat{\omega}\theta}$

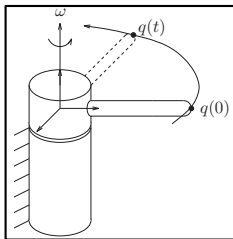


Figure 2.5

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- The *Exponential Map*
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Rodrigues' formula ($\|\omega\| = 1$):

$$e^{\hat{\omega}\theta} = I + \hat{\omega} \sin \theta + \hat{\omega}^2 (1 - \cos \theta)$$

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Rodrigues' formula ($\|\omega\| = 1$):

$$e^{\hat{\omega}\theta} = I + \hat{\omega} \sin \theta + \hat{\omega}^2(1 - \cos \theta)$$

Proof :

Let $a \in \mathbb{R}^3$, write

$$a = \omega\theta, \omega = \frac{a}{\|a\|} \text{ (or } \|\omega\| = 1), \text{ and } \theta = \|a\|$$

$$e^{\hat{\omega}\theta} = I + \hat{\omega}\theta + \frac{(\hat{\omega}\theta)^2}{2!} + \frac{(\hat{\omega}\theta)^3}{3!} + \dots$$

As

$$\hat{a}^2 = aa^T - \|a\|^2 I, \hat{a}^3 = -\|a\|^2 \hat{a}$$

we have:

$$\begin{aligned} e^{\hat{\omega}\theta} &= I + \left(\theta - \frac{\theta^3}{3!} + \frac{\theta^5}{5!} - \dots\right)\hat{\omega} + \left(\frac{\theta^2}{2!} - \frac{\theta^4}{4!} + \dots\right)\hat{\omega}^2 \\ &= I + \hat{\omega} \sin \theta + \hat{\omega}^2(1 - \cos \theta) \end{aligned}$$



2.2 Rotational Motion in \mathbb{R}^3

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Rodrigues' formula for $\|\omega\| \neq 1$:

$$e^{\hat{\omega}\theta} = I + \frac{\hat{\omega}}{\|\omega\|} \sin \|\omega\|\theta + \frac{\hat{\omega}^2}{\|\omega\|^2} (1 - \cos \|\omega\|\theta)$$

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Rodrigues' formula for $\|\omega\| \neq 1$:

$$e^{\hat{\omega}\theta} = I + \frac{\hat{\omega}}{\|\omega\|} \sin \|\omega\|\theta + \frac{\hat{\omega}^2}{\|\omega\|^2} (1 - \cos \|\omega\|\theta)$$

Proof for Property 3:

Let $R \triangleq e^{\hat{\omega}\theta}$, then:

$$\begin{aligned} (e^{\hat{\omega}\theta})^{-1} &= e^{-\hat{\omega}\theta} = e^{\hat{\omega}^T\theta} = (e^{\hat{\omega}\theta})^T \\ \Rightarrow R^{-1} &= R^T \Rightarrow R^T R = I \Rightarrow \det R = \pm 1 \end{aligned}$$

From $\det \exp(0) = 1$, and the continuity of \det function w.r.t. θ , we have $\det e^{\hat{\omega}\theta} = 1, \forall \theta \in \mathbb{R}$ □

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Property 4: The exponential map is onto.

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Property 4: The exponential map is onto.

Proof :

Given $R \in SO(3)$, to show $\exists \omega \in \mathbb{R}^3$, $\|\omega\| = 1$ and θ s.t. $R = e^{\hat{\omega}\theta}$

Let

$$R = \begin{bmatrix} r_{11} & r_{12} & r_{13} \\ r_{21} & r_{22} & r_{23} \\ r_{31} & r_{32} & r_{33} \end{bmatrix}$$

and

$$v_\theta = 1 - \cos \theta, c_\theta = \cos \theta, s_\theta = \sin \theta$$

By Rodrigues' formula $e^{\hat{\omega}\theta} = I + \hat{\omega} \sin \theta + \hat{\omega}^2 (1 - \cos \theta)$

$$e^{\hat{\omega}\theta} = \begin{bmatrix} \omega_1^2 v_\theta + c_\theta & \omega_1 \omega_2 v_\theta - \omega_3 s_\theta & \omega_1 \omega_3 v_\theta + \omega_2 s_\theta \\ \omega_1 \omega_2 v_\theta + \omega_3 s_\theta & \omega_2^2 v_\theta + c_\theta & \omega_2 \omega_3 v_\theta - \omega_1 s_\theta \\ \omega_1 \omega_3 v_\theta - \omega_2 s_\theta & \omega_2 \omega_3 v_\theta + \omega_1 s_\theta & \omega_3^2 v_\theta + c_\theta \end{bmatrix}$$

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2.2 Rotational Motion in \mathbb{R}^3

Taking the trace of both sides,

$$\text{tr}(R) = r_{11} + r_{22} + r_{33} = 1 + 2 \cos \theta = \sum_{i=1}^3 \lambda_i$$

where λ_i is the eigenvalue of R , $i = 1, 2, 3$

Case 1: $\text{tr}(R) = 3$ or $R = I$, $\theta = 0 \Rightarrow \omega\theta = 0$

2.2 Rotational Motion in \mathbb{R}^3

Taking the trace of both sides,

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Case 1: $\text{tr}(R) = 3$ or $R = I$, $\theta = 0 \Rightarrow \omega \theta = 0$

Case 2: $-1 < \text{tr}(R) < 3$,

$$\theta = \arccos \frac{\text{tr}(R) - 1}{2} \Rightarrow \omega = \frac{1}{2s_\theta} \begin{bmatrix} r_{32} - r_{23} \\ r_{13} - r_{31} \\ r_{21} - r_{12} \end{bmatrix}$$

2.2 Rotational Motion in \mathbb{R}^3

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Case 3: $\text{tr}(R) = -1 \Rightarrow \cos \theta = -1 \Rightarrow \theta = \pm \pi$

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2.2 Rotational Motion in \mathbb{R}^3

Following are 3 possibilities:

$$R = \begin{bmatrix} 1 & 0 & 0 \\ 0 & -1 & 0 \\ 0 & 0 & -1 \end{bmatrix} \Rightarrow \omega = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix},$$

$$R = \begin{bmatrix} -1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & -1 \end{bmatrix} \Rightarrow \omega = \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix},$$

$$R = \begin{bmatrix} -1 & 0 & 0 \\ 0 & -1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \Rightarrow \omega = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

Note that if $\omega\theta$ is a solution, then $\omega(\theta \pm n\pi)$, $n = 0, \pm 1, \pm 2, \dots$ is also a solution. □

2.2 Rotational Motion in \mathbb{R}^3 **Definition: Exponential coordinate**

$\omega\theta \in \mathbb{R}^3$, with $e^{\hat{\omega}\theta} = R$ is called the exponential coordinates of R

Exp :

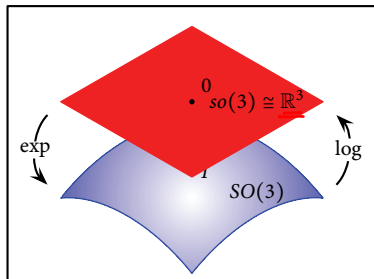


Figure 2.6

2.2 Rotational Motion in \mathbb{R}^3 **Definition: Exponential coordinate**

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Exp :

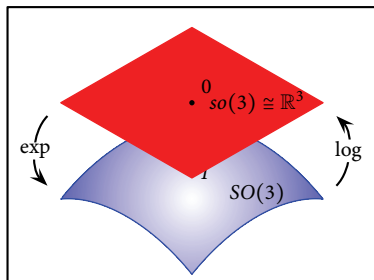


Figure 2.6

Property 5: \exp is 1-1 when restricted to an open ball in \mathbb{R}^3 of radius π .

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Theorem 1 (Euler):

Any orientation is equivalent to a rotation about a fixed axis $\omega \in \mathbb{R}^3$ through an angle $\theta \in [-\pi, \pi]$.



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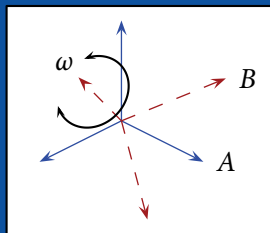


Figure 2.7

$SO(3)$ can be visualized as a solid ball of radius π .

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□ Other Parametrizations of $SO(3)$:

- XYZ fixed angles (or Roll-Pitch-Yaw angle)

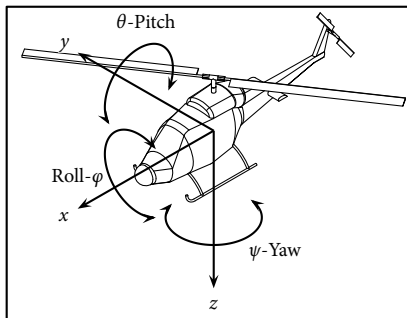


Figure 2.8

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2.2 Rotational Motion in \mathbb{R}^3

- XYZ fixed angles (or Roll-Pitch-Yaw angle) Continued

$$R_x(\varphi) := e^{\hat{x}\varphi} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \varphi & -\sin \varphi \\ 0 & \sin \varphi & \cos \varphi \end{bmatrix}$$

$$R_y(\theta) := e^{\hat{y}\theta} = \begin{bmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{bmatrix}$$

$$R_z(\psi) := e^{\hat{z}\psi} = \begin{bmatrix} \cos \psi & -\sin \psi & 0 \\ \sin \psi & \cos \psi & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$R_{ab} = R_x(\varphi)R_y(\theta)R_z(\psi)$$

$$= \begin{bmatrix} c_\theta c_\psi & -c_\theta s_\psi & s_\theta \\ s_\varphi s_\theta c_\psi + c_\varphi s_\psi & -s_\varphi s_\theta s_\psi + c_\varphi c_\psi & -s_\varphi c_\theta \\ -c_\varphi s_\theta c_\psi + s_\varphi s_\psi & c_\varphi s_\theta s_\psi + s_\varphi c_\psi & c_\varphi c_\theta \end{bmatrix}$$

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■ ZYX Euler angle

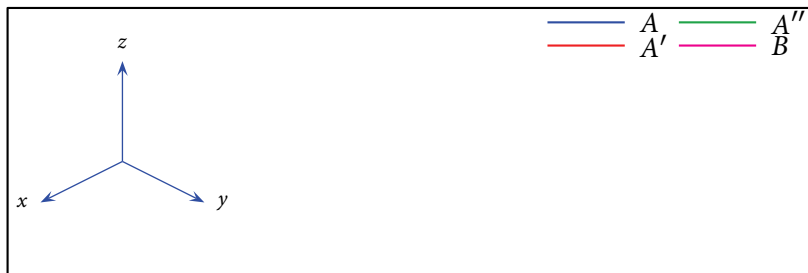


Figure 2.9

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■ ZYX Euler angle

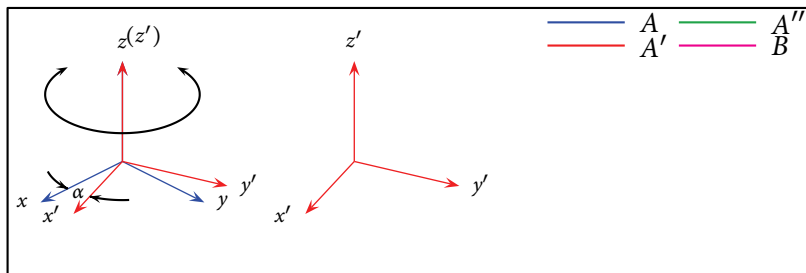


Figure 2.9

$$R_{aa'} = R_z(\alpha)$$

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■ ZYX Euler angle

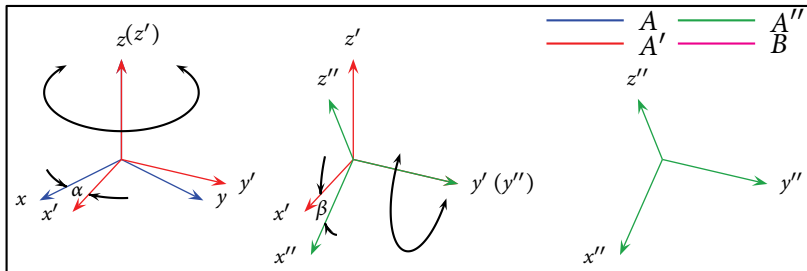


Figure 2.9

$$R_{aa'} = R_z(\alpha)$$

$$R_{a'a''} = R_{y'}(\beta)$$

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■ ZYX Euler angle

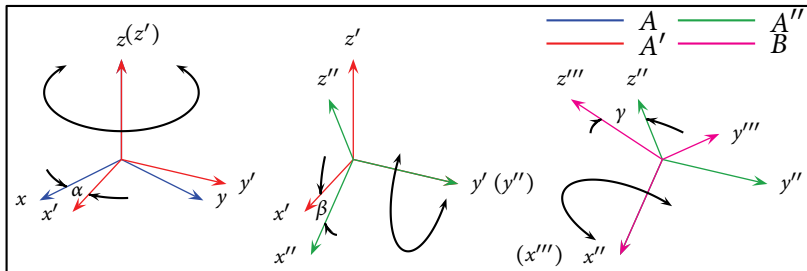


Figure 2.9

$$R_{aa'} = R_z(\alpha)$$

$$R_{a'a''} = R_{y'}(\beta)$$

$$R_{a''b} = R_{x''}(\gamma)$$

$$R_{ab} = R_z(\alpha)R_{y'}(\beta)R_{x''}(\gamma)$$

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- ZYX Euler angle (continued)

$$R_{ab}(\alpha, \beta, \gamma) = \begin{bmatrix} c_\alpha c_\beta & -s_\alpha c_\gamma + c_\alpha s_\beta s_\gamma & s_\alpha s_\gamma + c_\alpha s_\beta c_\gamma \\ s_\alpha c_\beta & c_\alpha c_\gamma + s_\alpha s_\beta s_\gamma & -c_\alpha s_\gamma + s_\alpha s_\beta c_\gamma \\ -s_\beta & c_\beta s_\gamma & c_\beta c_\gamma \end{bmatrix}$$

Note: When $\beta = \frac{\pi}{2}$, $\cos \beta = 0$, $\alpha + \gamma = \text{const} \Rightarrow \text{singularity!}$

$$\beta = \text{atan2}(-r_{31}, \sqrt{r_{32}^2 + r_{33}^2})$$

$$\alpha = \text{atan2}(r_{21}/c_\beta, r_{11}/c_\beta)$$

$$\gamma = \text{atan2}(r_{32}/c_\beta, r_{33}/c_\beta)$$

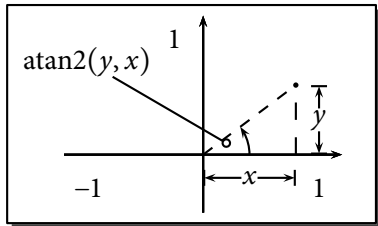
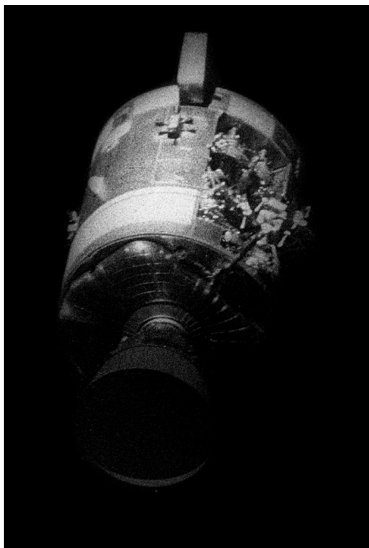
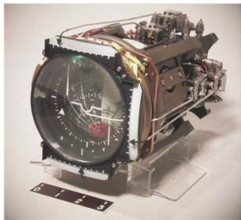
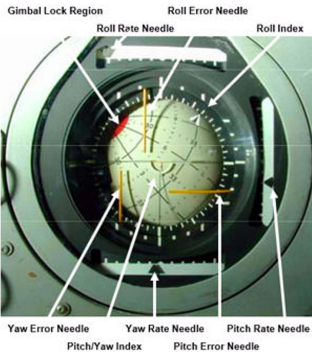


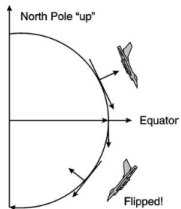
Figure 2.10

Apollo 10





F-16 Fly-By-Wire Fighter Jet



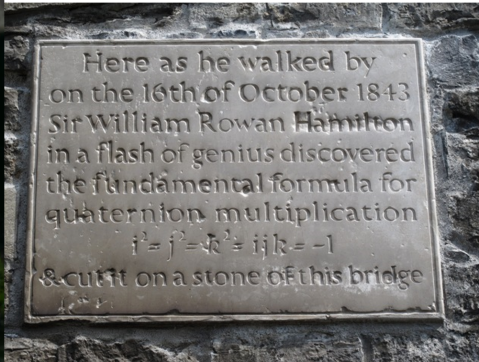
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Quaternions to the Rescue

Hamilton's Walk



2.2 Rotational Motion in \mathbb{R}^3

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§ Quaternions:

$$Q = q_0 + q_1i + q_2j + q_3k$$

where $i^2 = j^2 = k^2 = -1, i \cdot j = k, j \cdot k = i, k \cdot i = j$

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where $i^2 = j^2 = k^2 = -1, i \cdot j = k, j \cdot k = i, k \cdot i = j$

Property 1: Define $Q^* = (q_0, q)^* = (q_0, -q), q_0 \in \mathbb{R}, q \in \mathbb{R}^3$
 $\|Q\|^2 = QQ^* = q_0^2 + q_1^2 + q_2^2 + q_3^2$

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$$\|Q\|^2 = QQ^* = q_0^2 + q_1^2 + q_2^2 + q_3^2$$

Property 2: $Q = (q_0, q), P = (p_0, p)$

$$QP = (q_0p_0 - q \cdot p, q_0p + p_0q + q \times p)$$

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$$Q = q_0 + q_1i + q_2j + q_3k$$

where $i^2 = j^2 = k^2 = -1, i \cdot j = k, j \cdot k = i, k \cdot i = j$

Property 1: Define $Q^* = (q_0, q)^* = (q_0, -q), q_0 \in \mathbb{R}, q \in \mathbb{R}^3$
 $\|Q\|^2 = QQ^* = q_0^2 + q_1^2 + q_2^2 + q_3^2$

Property 2: $Q = (q_0, q), P = (p_0, p)$
 $QP = (q_0p_0 - q \cdot p, q_0p + p_0q + q \times p)$

Property 3: (a) The set of unit quaternions forms a group
 (b) If $R = e^{\hat{\omega}\theta}$, then $Q = (\cos \frac{\theta}{2}, \omega \sin \frac{\theta}{2})$
 (c) Q acts on $x \in \mathbb{R}^3$ by $QQXQ^*$, where $X = (0, x)$

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2.2 Rotational Motion in \mathbb{R}^3 □ **Unit Quaternions:**

Given $Q = (q_0, q)$, $q_0 \in \mathbb{R}$, $q \in \mathbb{R}^3$, the vector part of QXQ^* is given by $R(Q)x$, recall that

$$q_0 = \cos \frac{\theta}{2}, q = \omega \sin \frac{\theta}{2}$$

and the Rodrigues' formula:

$$e^{\hat{\omega}\theta} = I + \hat{\omega} \sin \theta + \hat{\omega}^2 (1 - \cos \theta)$$

then

$$\begin{aligned} R(Q) &= I + 2q_0\hat{q} + 2\hat{q}^2 \\ &= \begin{bmatrix} 1 - 2(q_2^2 + q_3^2) & -2q_0q_3 + 2q_1q_2 & 2q_0q_2 + 2q_1q_3 \\ 2q_0q_3 + 2q_1q_2 & 1 - 2(q_1^2 + q_3^2) & -2q_0q_1 + 2q_2q_3 \\ -2q_0q_2 + 2q_1q_3 & 2q_0q_1 + 2q_2q_3 & 1 - 2(q_1^2 + q_2^2) \end{bmatrix} \end{aligned}$$

where $\|Q\| \triangleq q_0^2 + q_1^2 + q_2^2 + q_3^2 = 1$

(continues next slide)

2.2 Rotational Motion in \mathbb{R}^3 □ **Quaternions (continued):**

Conversion from Roll-Pitch-Yaw angle to unit quaternions:

$$Q = \left(\cos \frac{\varphi}{2}, x \sin \frac{\varphi}{2}\right) \left(\cos \frac{\theta}{2}, y \sin \frac{\theta}{2}\right) \left(\cos \frac{\psi}{2}, z \sin \frac{\psi}{2}\right) \Rightarrow$$

$$q_0 = \cos \frac{\varphi}{2} \cos \frac{\theta}{2} \cos \frac{\psi}{2} - \sin \frac{\varphi}{2} \sin \frac{\theta}{2} \sin \frac{\psi}{2}$$

$$q = \begin{bmatrix} \cos \frac{\varphi}{2} \sin \frac{\theta}{2} \sin \frac{\psi}{2} + \sin \frac{\varphi}{2} \cos \frac{\theta}{2} \cos \frac{\psi}{2} \\ \cos \frac{\varphi}{2} \sin \frac{\theta}{2} \cos \frac{\psi}{2} - \sin \frac{\varphi}{2} \cos \frac{\theta}{2} \sin \frac{\psi}{2} \\ \cos \frac{\varphi}{2} \cos \frac{\theta}{2} \sin \frac{\psi}{2} + \sin \frac{\varphi}{2} \sin \frac{\theta}{2} \cos \frac{\psi}{2} \end{bmatrix}$$

Conversion from unit quaternions to roll-pitch-yaw angles (?)

† End of Section †

Today

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 - The *Exponential Map*
 - *Rodrigues Formula*
 - *Euler Angles*
 - *Quaternions*